

# Dynamic Systems - State Space Control

## - Lecture 14 \*

October 23, 2012

### 1 Equilibrium Points

**THEOREM:** A necessary and sufficient condition for the equilibrium point of  $\dot{x} = Ax + b$  is that the Eigenvalues of A have negative real parts. That is to say that the eigenvalues lie in the left half complex planes.

$$\dot{x} = Ax + Bu$$

$$y = Cx$$

State feedback  $u(t) = Kx(t)$

Compared to output feedback  $u(t) = Ky(t)$

The closed loop system with state feedback is:

$$\dot{x} = (A + Bk)x$$

Examples:

#### 1. RLC Circuit

Insert drawing here

Kirchoff's Law (Voltage Drops)

$$V = IR$$

$$V = dFL/dt$$

$$V = 1/C \int Ids$$

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\*This work is being done by various members of the class of 2012

$$L = d^2I/dt^2 + RdI/dt + 1/cI = E(t)$$

RLC circuit is controlled by voltage flux

## 2. Positioning a Mass Spring Dashpot System

Insert drawing here

$$m\ddot{x} + c\dot{x} + kx = u(t) (c > 0)$$

First orderizing the system yields

$$x_1 = x, x_2 = \dot{x}$$

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} 0 & 1 \\ -k/m & -c/m \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix}$$

The characteristic polynomial is

$$\lambda^2 + c\lambda/m + k/m = 0$$

$$\text{Roots } \lambda_i = -c/2m \pm 1/2m\sqrt{c^2 - 4km}$$

$$x(t) = a_1e^{\lambda_1 t} + a_2e^{\lambda_2 t}$$

If there is a large damping  $c^2 \gg 4km$

Insert drawing here

If either k or m or both are large

$$x(t) = a_1e^{-c/2m^*t} \cos(1/2m\sqrt{|c^2 - 4km|}t) + a_2e^{-c/2m} \sin(1/2m\sqrt{|c^2 - 4km|}t)$$

What can be achieved with feedback

$$u(t) = -k_p x_1(t) - k_v x_2(t)$$

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} 0 & 1 \\ -k/m & -c/m \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} + \begin{pmatrix} 0 \\ -k_p x_1(t) - k_v x_2(t) \end{pmatrix}$$

$$= \begin{pmatrix} 0 & 1 \\ -k/m - k_p & -c/m - k_v \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix}$$

The closed loop characteristic polynomial equation is

$$\lambda^2 + (c/m + k_v)\lambda + k/m + k_p = 0$$

Second order control systems

$$\ddot{x} + 2\zeta\omega\dot{x} + \omega^2x = 0, \zeta > 0$$

$\zeta$  is the damping ratio

$\omega$  is the natural frequency

The three cases of interest are  $\zeta < 1, \zeta = 1, \zeta > 1$

Why these are of interest

$$s^2 + 2\zeta\omega s + \omega^2 = 0$$

$$s = -\zeta\omega \pm \omega\sqrt{\zeta^2 - 1}$$

Case  $\zeta < 1$

Let  $a = -\zeta\omega, \omega > 0$

Then  $a^2 = \zeta^2\omega^2 < \omega^2$

Hence we can choose  $b$  s.t  $a^2 + b^2 = \omega^2$

The differential equation becomes:

$$\ddot{x} - 2a\dot{x} + (a^2 + b^2)x = 0$$

First orderize as follows

$$x_1 = x$$

$$x_2 = \dot{x} - ax/b$$

Then

$$\dot{x}_1 = ax_1 + bx_2$$

$$\dot{x}_2 = -bx_1 + ax_2$$

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} a & b \\ -b & a \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix}$$

We have computed

$$\begin{aligned} \Phi(t, 0) &= e^{\begin{pmatrix} a & b \\ -b & a \end{pmatrix} t} \\ &= \begin{pmatrix} e^{at}\cos(bt) & e^{at}\sin(bt) \\ -e^{at}\sin(bt) & e^{at}\cos(bt) \end{pmatrix} \end{aligned}$$

$$x(t) = x_1(t) = e^{at} \cos(bt)x_1(0) + e^{at} \sin(bt)x_2(0)$$

Insert drawing here

$$\begin{aligned} &= C e^{(at)} \sin(bt + \phi) \\ C &= \sqrt{x_1(0)^2 + x_2(0)^2} \\ \Phi &= \arctan(x_1(0), x_2(0)) \end{aligned}$$

If  $a > 0$

Insert drawing here

Case  $\zeta > 1$

$$\begin{aligned} \lambda_1 &= -\zeta \omega - \omega \sqrt{\zeta^2 - 1} \\ \lambda_2 &= -\zeta \omega + \omega \sqrt{\zeta^2 - 1} \end{aligned}$$

These roots are either both positive or both negative. Again let  $a = -\zeta\omega, \omega > 0$ . Then  $a^2 = \zeta^2\omega^2 > \omega^2$ . Hence there is a  $b^2 < a^2$  such that  $a^2 - b^2 = \omega^2$ . First orderize as the system

$$\begin{aligned} \dot{x}_1 &= ax_1 + bx_2 \\ \dot{x}_2 &= bx_1 + ax_2 \\ \begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} &= \begin{pmatrix} a & b \\ b & a \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} \end{aligned}$$

The transition matrix is

$$\begin{aligned} &e^{\begin{pmatrix} a & 0 \\ 0 & a \end{pmatrix} t} e^{\begin{pmatrix} 0 & b \\ b & 0 \end{pmatrix} t} \\ &e^{\begin{pmatrix} a & 0 \\ 0 & a \end{pmatrix} t} = \begin{pmatrix} e^{at} & 0 \\ 0 & e^{at} \end{pmatrix} \\ &e^{\begin{pmatrix} 0 & b \\ b & 0 \end{pmatrix} t} = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} + \begin{pmatrix} 0 & b \\ b & 0 \end{pmatrix} + 1/2 \begin{pmatrix} b^2 & 0 \\ 0 & b^2 \end{pmatrix} \\ &\quad + \dots \end{aligned}$$

The even terms sum to

$$\begin{pmatrix} e^{bt} + e^{-bt}/2 & 0 \\ 0 & e^{bt} + e^{-bt}/2 \end{pmatrix}$$

The odd terms sum to

$$\begin{pmatrix} 0 & e^{bt} - e^{-bt}/2 \\ e^{bt} - e^{-bt}/2 & 0 \end{pmatrix}$$

The transition matrix is

$$\begin{pmatrix} e^{at} & 0 \\ 0 & e^{at} \end{pmatrix} \begin{pmatrix} e^{bt} + e^{-bt}/2 & e^{bt} - e^{-bt}/2 \\ e^{bt} - e^{-bt}/2 & e^{bt} + e^{-bt}/2 \end{pmatrix}$$

Assuming (without loss of generality)  $b > 0$  we have  $|a| > b$

$$x(t) = C_1 e^{(a+b)t} + C_2 e^{(a-b)t}$$

$a < 0$

Insert drawing here

$a > 0$

Insert drawing here

Case  $\zeta = 1$

$$\dot{x}_1 = x$$

$$\dot{x}_2 = \dot{x}_1 + \omega x_1$$

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} -\omega & 1 \\ 0 & -\omega \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix}$$

$$x(t) = e^{-\omega t} (x_1 \cos t + x_2(0))$$

Summary -

Pole locations and closed loop dynamics. Poles occur as complex conjugate pairs.

Insert drawing here

## 2 Mass Spring System Reprise

$$m\ddot{x} + c\dot{x} + km = u(t)$$

Assuming we can measure  $x, \dot{x}$  and feed them back

$$u(t) = -k_p x(t) - k_v \dot{x}(t)$$

The closed loop system is

$$m\ddot{x} + (c + kv)\dot{x} + (k - k_p)x = 0$$

$$k + k_p/m = \omega^2$$

$$c + k_v/m = 2\zeta\omega$$

$$\zeta = c + k_v/2\sqrt{k + k_p}\sqrt{m}$$

When  $\zeta < 1$

$$\omega^2 = a^2 + b^2$$

$$\zeta = -a/\omega = -a/\sqrt{a^2 + b^2}$$

When  $\zeta > 1$

$$\omega^2 = a^2 - b^2$$

$$\zeta = -a/\omega = -a/\sqrt{a^2 - b^2}$$

When  $\zeta = 1$

$$\omega^2 = a^2 + b^2$$

$$x(t) = C_1e^{(a+b)t} + C_2e^{(a-b)t}, a < 0, b > 0, |a| > b$$

If  $-a$  is very large compared to  $b$   $\zeta = 1$

$$x(t) = Ce^{at}$$

Insert drawing here

If  $kv$  is very large  $\zeta \gg 1$  and  $-a$ ,  $b$  have similar magnitude

$$a + b = -\omega < 0$$

$$x(t) = C_1e^{-\omega t} + C_2e^{(-2b-\omega)t}$$

Insert drawing here

When  $\zeta = 1$  there is fast damping but there is danger if there are modeling errors of the system actually being underdamped

### 3 Roth, Hurwitz Asymptotic Stability

->  $G(s)$  ->

Suppose  $g(s)$  is a proper rational transfer function,  $g(s) = n(s)/d(s)$

Asymptotic stability depends on the zeros of  $g(s)$

$$d(s) = s^n + a_{n-1}s^{n-1} + \dots + a_0$$

They need to be in the left half plane.

$d(s)$  is said to be a Hurwitz polynomial when they are.

Since  $d(s)$  is the product of factors of the form  $s+a$  and  $s^2 + b_1s + b_0$

With  $a, b_1, b_0$  real and positive, a necessary condition for  $d(s)$  to be Hurwitz is that all coefficients  $a_k$  be positive. This however is not sufficient.

Exercise 1. For a polynomial having all coefficients positive such that it is not Hurwitz.

The Routh table associated with  $d(s) = s^n + a_{n-1}s^{n-1} + \dots + a_0$

$a_0$	$a_2$	$a_4$	$a_6$	...
$a_1$	$a_3$	$a_5$	$a_7$	...
$b_1$	$b_2$	$b_3$	$b_4$	...
$c_1$	$c_2$	$c_3$	$c_4$	...
$d_1$	$d_2$	$d_3$	$d_4$	...

The entries below the two rows of a's are"

$$b_1 = a_1a_2 - a_0a_3/a_1$$

$$b_2 = a_1a_4 - a_0a_5/a_1$$

$$b_3 = a_1a_6 - a_0a_7/a_1$$

$$c_1 = b_1a_3 - b_2a_1/b_1$$

$$c_2 = b_1a_5 - b_3a_1/b_1$$

$$d_1 = c_1b_2 - b_1c_2/c_1$$

etc ...

**ROUTH-HURWITZ THEOREM:** The number of sign changes in the left-hand column as you go down is equal to the number of zeros in the right half plane.

Example :  $d(s) = s^4 + s^3 + s^2 + 11s + 10$

Routh Table :

10	1	1
11	1	0
1/11	1	0
-120	0	0
1	0	0

$$b_1 = 11 * 1 - 1 * 10/11$$

$$b_2 = 1 * 1 - 1 * 0/11$$

$$c_1 = 1/11 - 1 * 11/1/11 = -120$$

Two sign changes in the first column means two zeros in the right half plane.

Reference : Lerrant, Lepschy, Viavo. 1999. "A Simple Proof of the Routh Test". IEEE Transactions on Automatic Control. 44(6). 1306-1309.